

AVIATION

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VOLUME XIII

Number 8

SPECIAL FEATURES

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DESIGNING LANDING GEAR SHOCK ABSORBERS
DETROIT AERO CONGRESS
GORDON BENNETT BALLOON RACE

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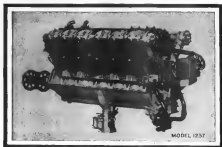
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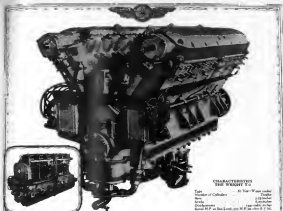
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Displacement	1,200 cu. in.
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AUGUST 21, 1922

No. 8

Responsibility for Airplane Accidents

THE publicity given one airplane accident by newspapers is usually many times the space given to any successful achievement of aircraft; and in just the same manner in the mind of the public afterward. As practically all the accidents that happen to airplanes might have been prevented if it is a general point to determine who or what is responsible for the construction of these machines.

Inspection and regulation are the two imperative needs of commercial aviation today. Federal legislation is the only remedy. And this necessary legislation is being held back because the Congressmen at Washington will not get together on some sort of bill to regulate flying. The Wadsworth bill for the specific purpose passed the Senate last February. It is now passing provisionally in the Committee of Interstate and Foreign Commerce awaiting the turn to come up for hearing. Meanwhile deaths come from the use of machines that never should take the air and from the misuse of airplanes through ignorance or carelessness.

Until some law is passed placing the control of flying somewhere there will continue to be accidents, and the responsibility will be due to the delay in passing federal legislation. Every official who blocks progress will share the responsibility for these accidents. Clarence Wadsworth, in whose committee the Wadsworth bill now rests, has a great responsibility and also a great opportunity.

Get Back to Fundamentals

IT is probably a known feeling, in engineering as well as in politics, to meet upon such words. It is the lay man's way to avoid making a thing out to its fundamental principles, the ignorant man's substitute for real knowledge, the prejudiced man's expedient, to satisfy his own conscience. There is no such thing as intrinsic moral as some particular form of government, without reference to the people who impose it and those whom it is to serve. There is likewise no way of judging the merit of some particular construction without reference to the materials used and the purpose intended.

The above may sound very philosophical, but it is reasonable how much sense in aeronautics seems to be put on such things. It is as if legislation should discuss the relative merits of the tailfin, the suspension, or plate prior type of construction without reference to what it is to be made of or where it is to be used. Yet the same men who would ridicule such a controversy are prone to speak of the lightness, or the relatively broad wingspan, the aerodynamic or the rigid type of airplane as if the same words were large words with some such power for good or evil.

The fundamental question is not "What type is it?" but

"What are you going to do with it?" With this as a criterion, and sound engineering as a working basis, the designer will do well to forget about the "type" till the job is done. Finding a name for it is a mere detail.

Why the Mooring Mast?

THE airplane mooring mast serves at least one useful purpose, that of keeping the ship headed approximately into the wind. But are modern airport engineers so lacking in resources that they have to swing the ship from the tip of a tail mast or tower to achieve this simple result? We must admit that a mast would be almost necessary for an emergency mooring over rough country or in the heart of a city. But most of the more difficult moorings are planned to be located at some regular air station having a large flat field available. In a case like this there would seem to be greater advantage in a low mooring, practically on the ground.

One often hears the argument that "proper freedom of motion must be allowed in the ship's vertical angle as well as direction directionally". But for what purpose? There is not enough vertical component in the wind near the ground to cause serious aerodynamic forces, except indirectly due to shock. But shock is possible only when there is motion. Hence the vertical motion allowed at a mooring mast is the cause of the very thing it seeks to control. Why not get the ship down as close to the ground as possible and hold it there rigidly, as far as vertical motion is concerned?

Who's Who in American Aeronautics

THE appreciative comment that has been received about the first edition of "Who's Who in American Aeronautics" has been extremely gratifying. The omission of some of these biographies that should have been included, could they have been secured, was inevitable but every effort was made to make them available.

The second edition which will be published early in 1923 should contain many additional sketches of men who are actually interested in flying in the United States. Every pilot is invited to send a sketch of his career in aviation "Who's Who" form. It is impossible to secure addresses in all cases so this opportunity is taken to reach those who would otherwise be overlooked.

It is too early to establish definite rules for inclusion in the book. Everyone who fits should have available a reference work which registers him with others who are in the industry or interested in commercial aviation or the sport of flying.

You will be assisting the general good of aeronautics in the United States by sending in your biography.

New Japanese Military Airplanes

Interesting Particulars of Four Military Ships of Japanese Manufacture and Design



Fig. 1. Aichi biplane. Fig. 2. Mitsubishi single-engine pursuit machine. Fig. 3. Kawasaki 3-engine. Fig. 4. Shoroku diplane.

The construction of Japanese military aircraft appearing on this page, which have just reached this country from Japan, afford an interesting insight into the progress made by Japanese aircraft construction in the last year or so.

It is not so very long ago when pictures of Japanese airplanes showed them to be very primitive affairs indeed, and as it is hard to see this improving condition in Japanese pilots were frequently reported. That Japan has progressed this stage and that it is making rapid strides toward building up an efficient air force, manned by a skilled personnel and backed by an industry relying on national resources, is shown by various indications. During the past year numerous long distance flights have been made by Japanese air forces, and, in fact, the Imperial Army maneuvers were held with both parties disposing of up to date planes of observation, pursuit and bombing airplanes, and much attention was also devoted toward building up Japanese naval aviation.

The construction on the page are believed to be the first published in this country of up to date Japanese military aircraft of Japanese manufacture. While namely British, French and German, influence is still apparent in the design of these machines, they nevertheless disclose a certain originality which undoubtedly stamps them as the forerunners of really original Japanese aircraft.

While the information regarding the characteristics of the machines illustrated is incomplete, it reveals enough in connection with the photographs to obtain a fairly comprehensive understanding of the possibilities of the different types. The Aichi biplane illustrated in Fig. 1 is apparently a single-engine pursuit plane, showing some English influence and having for characteristics the following: Length 7.715 m., height 2.30 m., chord 1.535 m., total wing area 28 sq. m.,

220 hp. Maybach engine, total weight 1675 kg., high speed 180 km. p. h.

Fig. 2 illustrates the 1923 model single-engine fighter produced by the Mitsubishi Dockyards, which shows considerable foreign influence. Span 5.350 m., length 2.300 m., height 2.10 m., 300 hp. Hispano-Suiza engine, total weight 1675 kg. This machine is armed with two machine guns and is equipped with emergency flotation air bags.

The Kawasaki 3-engine biplane shown in Fig. 3, with 300 hp. Bata, a well known Japanese civil pilot standing in front, is constructed in many respects of the German war-time L.V.G. two-engine fighter and carries observation machine. Its characteristics are: Span 11.00 m., height 7.50 m., height 3.75 m., total wing area 31 sq. m., 300 hp. Maybach engine, high speed 180 km. p. h.

No particulars are at hand regarding the Shoroku biplane illustrated in Fig. 4. The caption on the photograph calls it "the most highly speed machine, civil aviation in Japan", but judging by its looks, reminiscent of the Spad XIII, the machine rather looks like a single-engine fighter.

Perfect Air Mail Record

The transcontinental air mail service has had a perfect record for the past three weeks according to reports received by the Post Office Department from the three operating airlines. All three airlines had 100 per cent performance, which means that every plane flew as scheduled and that every flight was completed. No plane was damaged by a bad landing or forced to undergo repairs.

Designing Landing Gear Shock Absorbers

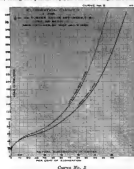
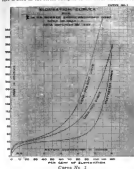
Practical Method for Determining the Size of Cord, Number of Loops, Tension, etc. for a Given Service

By Orrin E. Ross

There has been little research and fewer technical articles written on the design and construction of rubber shock absorbers for airplane landing gears than for any other department of aeronautical engineering. The reason is not that engineers and designers regard the landing gear as unimportant or as a secondary factor in the safety and comfort of the occupants of an airplane. Although considerable time and care is used in the stress analysis of the steel structure of the

gear, the only good quality about the old style "over and under" shock absorber is its lightness, which of course is an important factor in making an airplane perform airplane.

The primary object of this article is to deal with the actual design of a rubber shock absorber, but the reader takes the opportunity here to express his belief that all shock absorbers for racing and general ships, as well as the larger and heavier



chassis, yet when a shock absorber is needed it is generally passed at and then interested and tested. If the test is not satisfactory, the shock absorber is changed, either by using a different sized cord or by varying the initial tension or the number of wraps of the cord. On the other hand, when the test is satisfactory to the person making it, the shock absorber is used on the ship it is meant for, and, alas, it is used on all other ships of the same design without any account being taken of the variation in the quality of the elastic cord. This is where the trouble lies, as will be seen later.

The above method of design and construction is more popular on small airplanes weighing 2000 lb. or less. The reason is that the cord can be wrapped directly on the landing gear assembly. The method for larger machines is to have "barrel wrapped" shock absorbers. The former style has more than one disadvantage; it is next to impossible to figure the number of loops of the cord or the desired initial tension; the loops are not the same size and therefore the strands will not have the same elongation; the cord being wrapped on more than one layer produces chafing which affects the life of the shock absorber; also, it requires some time to change the cord than it would if the shock absorber was barrel wrapped and rigid

type, should be made so that the characteristics and shock absorbing qualities of the rubber, when assembled, are figured to a fair degree of accuracy.

Caution Engineers

Due to the differences in the elasticity and elongation of different lots of the same kind and size of rubber cord, it is believed that every one who has anything to do with the designing, wrapping, inspection or changing in any way of an airplane shock absorber should know the characteristics of the particular cord he is using and should know how to calculate the necessary conditions to insure the safety of the ship as well as that of the pilot and the passengers. In other words, it is found by experiment that no two batches of cord of the same size will give the same elongation curve when plotted load in pounds against elongation in per cent or inches.

Although this defect exists, it is nevertheless beyond the control of the manufacturers. When ordering, one may specify a cord which shall have 100 per cent elongation for a given range of loads, but, when the cord is assembled, it may be within the range or it may not. For instance, the United States government specifications for landing gear shock ab-

series cord requires the following loads for 100 per cent elongation—

Diameter of finished cord	Load in pounds
1/8"	100 to 125
3/16"	165 to 225
1/4"	215 to 300

Condensable experience with cord bought under this specification has shown that the cord requires a greater load than the loads include, and that the cost of the loads required for various per cents of elongation are such that the curve plotted to the results is too straight. This shows that in the manufacture of the cord, the refinery and shock absorbing qualities of the rubber have been sacrificed in order to meet the required load for 100 per cent elongation. In other case, it is necessary to know what the characteristic of the whole cord are in order intelligently to design a shock absorber. It is not enough simply to know what the loads are for 100 per cent elongation.

In order to make these points clear, we will solve a few problems in the actual design of the shock absorber. This method of calculation holds good for any size or weight of replace, but we will choose a fairly large one weighing, say 5000 lb. fully loaded.

Problem No. 1

We will assume that we are starting on a preliminary design. All we know is the weight of the ship, the points of attachment of struts and axle, the size of the wheel and the length of the link and the conditions for which we want to analyze our struts and determine the distance of safety. We know the style of shock absorber we intend to use, (a removable hook wrapped steel), but we do not know the number of loops nor the size of the cord we need. Neither do we know the distance between the shock absorber and the wheel which will give us the best elongation with an assumed movement of 4 in. at the shock absorber. We can arbitrarily use 3/16 in. diameter spindles and try 1/4 in. diameter cord to start with. We can assume that we have no cord on hand so we naturally will not know what kind of an elongation curve it will have. We can use a curve, however, that we had for cord on another ship and which we will call, here, Curve No. 1.

Although half the weight of the ship is only 2500 lb., due to the condenser action of the wheel and the resolution of vertical and horizontal forces which we will not solve here, we find that the vertical reaction at the shock absorber or the force acting on the rubber is 5000 lb.

Let us proceed with the calculations as follows—

Solving for the minimum distance between the spindles. See Fig. 1.

Date	Load—4000 lb.
Impact factor—4/5	
Ultimate load 32000 lb.	
3/16 diameter cord	
Elongation curve No. 1	
Assume 50 strands or 34 loops	

4000
— 72 lb. per strand initial tension

40
72 lb. gives 54 per cent elongation (2000 curve)

2000
— 320 lb. per strand maximum tension

120 lb. gives 106 per cent elongation.

Length of one loop with axle at rest = $1/4 + 2 + 2 = 5.5 + 2 = 7.5$

Length of one loop extended = $100 \times (5.5 + 2) = 8.5 + 2 = 10.5$

184
— 1.54

Length of one loop at maximum deflection of rubber = $1/4 + 2 + (2 + 4) = 1.25 + 2 = 3.25$

Elongation = $12.5 + 2 = 9.5 + 2 = 11.5$

1.54



Fig. 1

$$11.5 + 2 = (5.5 + 2) \times 1.54$$

$$1.54$$

$$= 165 \text{ per cent}$$

$$5.5 + 2 = 7.5$$

$$X = 9.33 \text{ in.}$$

New the measurements in this preliminary example may not be the same as some other design would choose. That a motor car may think 50 strands either too many or too few in either case, he could try 75 strands or 44 strands and find that the results were only slightly. Still another designer might think that a movement of 4 in. is not enough for the axle. But if he solves the problem in this case with a movement of 4 1/2 or 5 in., he will find that the results would be in such good luck that he might run into difficulties in construction or weights or strains (long possibilities). The size of the wheel and the length of the tread has a good deal to do with determining the size of the shock absorber because the ground appearance is a factor to be considered as well as its weight and load resistance.

Problem No. 2

Then let us make it as small as possible for all these reasons. We can arbitrarily take the distance between the spindles to be, say 8 1/2 in. Then we will have to assume an initial tension and solve for the impact factor or vice-versa. The method of computation in both cases are similar.

By decreasing the distance between the spindles, it is really seen that there will be less rubber in a loop if the same initial tension is maintained and that, therefore, there will be a greater elongation if the movement of the axle is constant.

Solving for the impact factor necessary to deflect the shock absorber to its maximum position (see Fig. 2).

Date: 40 strands or 34 loops.
Initial tension, assume 72 lb. per strand which is enough to support the weight of the ship at rest.
72 lb. gives 54 per cent elongation.

Curve No. 1.
Length of one loop with axle at rest = $1/4 + 2 + 2 = 5.5 + 2 = 7.5$

Length of one loop extended = $100 \times 7.5 = 750$

— 23. = 34.92 in.

104
Length of one loop at maximum elongation of rubber.
 $1/4 + 2 + 2 = 5.5 + 2 = 7.5$

31 = 14.92 = 30.98 = elongation 16.98

— 165 per cent cord gives 350 lb.

14.92
— 4.85 impact factor.

2000
— 4.85 impact factor.

40
New, suppose we design our spindles and strut structure on this basis and under one material. When the shock absorber cord is secured, we test it and run a curve which we will call Curve No. 2. Upon examination of the curve, it will be noticed that for the lower values of per cent there are relatively high load values, also, that the belly is not very pronounced, tending to make the curve a straight line. As the shock absorber qualities of a cord start between 20 per cent and 115 per cent elongation, it is necessary for it to have a good amount of resilience which keeps the curve out between these points. In short, the less elasticity is a substitute the straighter the elongation curve. Steel gives a straight line curve.

If we use this rubber, starting with an initial tension of 72 lb. per strand and solve for the impact factor by the method shown above, we find that the result is 2.5. An impact factor of 2.5 is not sufficient, so we begin to test about to

find a way to use this rubber. We try several greater values for the initial tension and still the impact factor is too small. We use 80 lb. per strand (initial tension which would render the shock absorber of little or no use in the ship in landing and which would cause undue and probably dangerous vibration throughout the whole quads. Still with the high initial tension, the impact factor is only 3.36.

There are only two things left to do. One is to release the shock absorber to test the cord, which would be a great expense if the drawings for the whole chassis were made and probably some of the parts already manufactured. The other is to order new cord. Let us assume that the latter is the way to do in this case.

Problem No. 3

The curve for this new batch of cord will be called Curve No. 3. It will be noticed that for an initial tension of 72 lb. per strand, we get an impact factor of 3.8. This is not as bad as Curve No. 2, but still it is not what we desire. However, good results are obtained by starting with the desired impact factor of 4.5 and solving for the initial tension in the following manner:

Date: Load 4000 lb.
Impact factor 4.5.
Ultimate load 32000 lb.
40 strands or 34 loops
Elongation curve No. 3
X = Initial tension in per cent.

20000
— 320 lb. per strand
220 lb. gives 118 per cent elongation.

See Fig. 3.
Length of one loop with axle at rest = 7.5 in.
Length of one loop at rest, elongation of rubber 31 in.

Fig. 2
Length of one loop extended = $X \times 31 \text{ or } X \times 31 \text{ in.}$

23 = 31
 $1 + X = 2.1$

$X = \frac{11.25}{31} = 36.33 \text{ per cent.}$

36.33
55.75 per cent gives 75 lb. per strand.

75 \times 45 = 3375 lb. This is more than the load on the wheel and is not conducive to the easiest riding qualities of the ship in landing, but it is not excessive and is the most profitable solution under the circumstances.

It may be stated here that the load condition for a new shock absorber is to wrap it with an initial tension slightly greater than enough to support the ship at rest and to have the greatest tension enough to give an impact factor slightly greater than the actual desired one. The reason for this is that the rubber, at first, suffers a slight depression or acquisition a permanent set which tends to loosen the rubber to its proper initial tension and also becomes its impact factor to the proper amount.

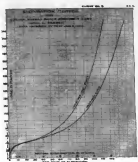
These elongation will demonstrate in the rubber that it is not enough to have the load required by a cord to produce 100 per cent elongation. We must know all the loads for all the per cents of elongation. Whether a man is an engineer, a doctor, a ship or field superintendent, a farmer or a workman, it is well for him to know all the characteristics of a rubber cord before wrapping a shock absorber. It is not enough to wrap a shock absorber with an initial tension that will support the ship when standing still and take a chance on the landing, which will extend when the axle moves to its maximum position. The cord may be such that the initial tension would have to be greater than that which will barely



Fig. 2

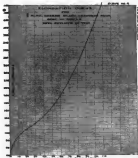


Fig. 3



Curve No. 3

support the ship in order to secure the maximum tension that will produce the proper impact factor. To study our argument a little further and to prove that different lots of the same kind of cord are not suitable for the same shock absorber, let us suppose that you are given in



Curve No. 4

completed and in the field. Suppose that the shock absorbers used jumping and that the store house at the field has only $\frac{1}{2}$ in. diameter cord. The strange machine would probably wrap the shock absorbers by going out as long as they reported the ship at rest without deflection it would take a chance on them absorbing the shocks properly. Instead, if he would take a pound and paper and figure it out, he would probably get results as follows—

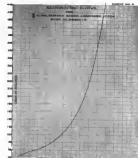
26 loops of $\frac{1}{2}$ in. cord occupy 17 in. of spindle

37

17 in. of $\frac{1}{2}$ in. cord would be — or 36 loops.

5/6

26 loops = 52 strands.



Curve No. 5

Curve No. 4 shows us that this cord has very little resistance and our figures will show that it is of little or no value for use shock absorber. If we solve for an amount factor by starting with an initial tension per strand of — or 9445

52

lb. which gives us elongation 22 per cent, we get a result of 1.88. Such an impact factor at all times would be considered. Again, if we start with an impact factor of 4.5 and solve for what our initial tension must be, we find that it is 178 lb. per strand. $178 \times 52 = 9256$ lb. This would be very similar to using jump for a shock absorber or fastening the side directly to the strands.

Consider that another reel of cord can be purchased from the store house which will give an elongation curve (the Curve No. 5). One ship would be flying in a short time. By starting with an initial tension sufficient to support the load and solving for the initial factor, we will get a result of 4.67, which is considered good.

Another thing, we can facilitate calculations to an approximate degree by calculating the total length of cord and not making up our mind before we begin. That is, the length of the cord for each shock absorber is the length of one loop, unstretched, times the number of loops, plus the amount it takes to go around the handles or loops at the ends.

International Air Convention

Fourteen states having on June 1, last, deposited the act of ratification of the Convention for the Regulation of Air Navigation with the French Minister of Foreign Affairs, the convention came into force with respect to these states on July 11. The ratifying states are the following: Belgium, Bolivia, British Empire (including Canada, Australia, New Zealand, South Africa, India, Persia, Greece, Japan, Portugal, Serbo-Croat-Slovene Kingdom and Spain). It is difficult to find the following states, which were not signatories to the Convention originally, have adhered to it: Panama, Peru, Liberia and Nicaragua. Consequently the International Convention is now in force in eighteen sovereign states at once.

In accordance with the terms of the L.A.C. the French government received on July 11 (forty days after the act of ratification) the International Convention for Air Navigation. This convention has the power to change by a three-quarter vote any provision of any of the statutes in the Convention. These statutes deal with the following subjects: (A) The Marking of Aircraft, (B) Certificates of Airworthiness, (C) Log Books, (D) Rules as to Lights and Signals, and Rules of the Air, (E) Qualifications Necessary for Piloting Fixed and Movable Aeroplanes, (F) International Aeronautical Maps and General Markings, (G) Meteorological Information, (H) Customs.

Lieut. Hinton on Flight to Brazil

Lieut. Walter Hinton and four companions left New York on the morning of Aug. 10 in the H323 Seagull. On their first flight to Rio de Janeiro, Brazil, to attend the Brazilian Centennial Exposition.

It is understood that the flight is being made under the auspices of the New York World and Union News. Lieut. Hinton reported to reach Santos, on Brazilian Island, S. C., a distance of 400 miles, on the first day of his journey.

Japanese Naval Aircraft



Above: Japanese "Seagull" torpedo plane (420 hp) by "Seagull" (400 hp). Below: Japanese "Seagull" (400 hp) by "Seagull" (400 hp). Below: Japanese "Seagull" (400 hp) by "Seagull" (400 hp).

Both machines are of British manufacture.

Preparing for the Detroit Aero Congress

Thirty-Nine Entries in Land and Water Speed Events Announced by Detroit Aviation Society

There have been many aviation conventions and flying meets held in the United States since the Wright brothers made the first flight in 1903, but the Detroit National Aero Congress and Exposition is to be held in Detroit in October gives every indication of being the most significant and widely expected aviation gathering yet held in the history of American aviation. Divided into two parts, the aero congress will consist of the annual and winter in Detroit and only the annual convention of all the various elements in aviation but the most respectable series of airplane races yet held in this country and, possibly, abroad. It is the expected nucleus of foreign governments to be shown in character to make a mark on the races, so that the meet is recognized as of international importance.

This, at least, was the assumption of Sidney D. Walden, president of the Detroit Aviation Society, when he sent in systems to all governments suggesting that they send their best flying machines and pilots to compete against the large field of American entries, entries by the way, which have already been officially entered. While the complete list of entries has not yet been announced, due to the number of letters awaiting to be answered, more than thirty of the latest designs yet built are being assembled in various aircraft factories throughout the country for their appearance several weeks before the meet. Typical among the letters received by the head of the Detroit Aviation Society, which is that from Secretary of War John W. Weeks. It reads:

My Dear Mr. Walden:

July 26, 1932

Mr. Sidney D. Walden, President, Detroit Aviation Society, Inc., 4015 Woodward avenue, Detroit, Michigan. My dear Mr. Walden:— "Referring to my letter of March 24, concerning the entry of the Army in the air meet to be held under the auspices of your Society, I am now pleased to inform you that the project has been definitely approved and the Chief of Air Service has been directed to communicate to you the events in which the Army will participate.

"I wish to take this occasion to express to you my appreciation of the work that your Society is doing in the interest of aerial activity and the wish that your coming meet will be successful in all of its varied projects.

Yours very truly,

(Signed) JOHN W. WEEKS,

Secretary of War.

More letters number in here and expressing the same kind of well-wished appreciation, have been received by Mr. Walden. It has been known for several years that the holding of an aviation meet demands not only the attention of a few interested individuals, but the control and protection, or supervision of the entire community. When enter manufacturers have appointed legitimate aviation enterprises, they have been rewarded. Where unscrupulous and public cooperation has failed or lagged, there will be found the record of failed efforts and consequent loss of public confidence in all aviation.

Five Principal Events

Five principal events are scheduled for the national airplane meet. The first will be held on Lake St. Clair this day, October 1, a few days in advance of the land meet which will be held at Bellefonte Field, Mt. Clemens, Mich. The chief event on Lake St. Clair will be the Curtiss Martin Flying Trophy Race. The distance to be covered is approximately 100 miles with eight laps around a closed, triangular course of 20 miles with the longest leg of the course paralleling the Detroit shore northwest from the starting point near Belle Isle. So far there are eleven entries for that race.

The Detroit Aviation Society has headquarters at 4015 Woodward avenue, and there the executive staff handles the entire convention, in addition to preparing to handle the aviation which will occupy Bellefonte Field Oct. 12, 13 and 14, the days of the over-land meet. Two of these are on the trophy, the Detroit News Annual Mod. Trophy race for large open-cockpit airplanes, and the Aviation Century Club at Detroit Trophy race for light commercial airplanes. On the thirteenth the Liberty Eagle-Benders' Trophy race for observation type (2 passenger) airplanes will be held. On the thirteenth and fourteenth will be held, and in these the First Powerst Group at Bellefonte Field will play no small part. Mac, Carl Smith's command at Bellefonte is the center for the advanced flying air force of the rotary Army. It is being built up rapidly and equipped with new planes which are arriving as predicted.

The Pulitzer Speed Chase

Of all the races the premier event will be the Pulitzer Trophy Race on Saturday, Oct. 15, when some of the fastest planes ever built will compete in what has become to be known as the American speed chase. There are twenty-eight entries in this race. Both the Army and Navy have entered three latest machines, a majority of which are now being constructed secretly in various aircraft plants. It is confidently predicted that more than one victory will be won by the Army, which is the same for all of the over-land races, starts from Bellefonte Field and extends to Troy, Greenham to Packard Field and thence back to Bellefonte, a triangular course of approximately 60 miles, which the Pulitzer contestants must fly four times, or 160 miles, non-stop.

The winner of last year's Pulitzer contest at Omaha, Neb. averaged 175.7 m.p.h. for the entire race. Many of the machines entered this year are expected to make in the neighborhood of 200 m.p.h. Such a speed means that the pilots will fly their ships at the rate of 300 ft. a second and because they must round each point in the course within 500 ft. of the ground, it is predicted persons they will occasionally have to leap about that much.

Other pilots expect to fly in the race from various cities. Many of them are delegates to the convention, which will convene at the Detroit Hotel, because for the first time in America, all the elements in aviation are endeavoring to organize a permanent national aeronautical association which shall represent the national thought and the ideas of all individuals—both commercially and for the national defense. The local men who have been interested in aviation for years endeavored last year to stand out public sentiment nationally, with a view toward conducting the 1933 aviation meet in Detroit. But there was insufficient support. The national aviation was thwarted. The government, protesting company which the times had forced upon it in all branches, was unwilling to offer support to a venture which it recognized must be supported by the entire aviation community. The entire aviation, lacking a credible market and with its government orders curtailed to a minimum, obviously could not contribute much to a flying meet. The Detroit Aviation Society, therefore postponed its meet until co-operation could be secured.

Thorough Preparation for Meet

Starting more than six weeks ago, the Detroit Aviation Society went East, West, South, sending public relations, national and local, and the state of the aeronautical industry. It was found that the American people are beginning to accept flying as a practical means of communication, travel and transport. The aviation industry is beginning to be recognized as a legitimate business. Legislation was being passed or

considered to regulate flying. Moreover the federal government had repeatedly asserted the importance of the airplane, its influence on our national life and above all the urgent need of developing aviation as a means of national defense. Our observers abroad had reported increasing activity on the part of foreign governments which through their air armaments, departments of war and marine, had set their best designers at work developing planes for every conceivable military and naval use.

There are expectations that Great Britain will send one or two planes to participate in the Detroit race, but it is rumored among the circles in Washington that if foreign governments do not enter their planes in the national race, it will be for the same reason and they do not want to display their latest designs here and thereby help stimulate American aircraft development.

Among the nations so far confirmed are twenty-eight different types of aircraft that they say are produced in quantity for any emergency as well as for peace. With three exceptions they have never before had an opportunity to match their performance with other craft. Practically every new type of aircraft of recent development will perform in the Detroit race.

Detroit Planning for the Future

It requires no further explanation to show the reason why Detroit's interest in the national airplane race. Detroit leads the lead in the motor industry because it is in Detroit that men of reason and confidence see the possibilities. A majority of these visualize the vast, limitless future of the airplane. They know that with its development will come factories, plants, shops, stores, services, and most of all, especially for aviation purposes. The city that prepares itself to accommodate the demand for three machines will be more quickly enabled itself on the way of the nation.

The Detroit Aviation Society is handling the local arrangements for the convention, its executives being the same as for the race. They include: Sidney D. Walcott, President; H. W. Allen, Vice-President; Stance P. Ramsey, Secretary; Wm. E. Mangan, Treasurer; Edward H. Emswore, Counsel.

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American Glider Successful

The success of Edward L. Allen of the Massachusetts Institute of Technology at Clermont Perrot in his American built glider planes in the front ranks of glider pilots.

The first day he flew 3 am. 18 am. on four flights, making complete mastery of control and maneuvering at the wheel. The next last effort was that of Switzerland, whose representative remained in the air for a total period of 1 am. 4 am.

The scientific results obtained by Allen's flights are extremely noteworthy in the field of aviation research. The expert of Prof. Edward P. Warren of the Massachusetts Institute of Technology. It is his opinion that the American glider, if equipped with a few horsepower motor, can take the air at any time and from any position, whether it be a hill, a lake, or a field. Well known pilots like Bonaparte, Paulhan, Cope, and Farber agreed with this opinion. All said that Allen's control of the machine was the best ever seen.

Allen is his performance, control, and maneuvering from the taking off point and took advantage of every part of wind. He did not attempt to make a record duration flight, as he was greatly concerned with the regular manner of the ground, take, and other experiments. He parked out most spots for landing. In some of his flights he lost control of the glider for a short time, but he was able to land.

Allen's flights, which were made in a house, varied from twenty to twenty-five miles an hour, were taken from the ground, and were made in a house. In one instance, the glider landed nearly fifty feet in altitude over the starting point, but the wind helped temporarily the pilot had to make a one day in order to get up speed.

The American machine was launched by a rope which was attached to a hook on a scaffold that the rope was automatically released when the machine left the ground. The pilot of the French machine has decided to adopt a similar launching device, as it is considered the most powerful. The American machine requiring only fifteen to twenty feet on the ground before starting, while Bonaparte in his attempts yesterday needed more than fifty feet.

The experts who have been on the field since the opening of the contest are in agreement that the standard type of glider built along the lines of the lighter airplanes, with increased lifting capacity and reduced weight, have had the better of the competition over the "heavy" machines. None of these latter has succeeded in taking off the "soft," the "flying field" and the "boppers" remaining on the ground unable to make any but the shortest hops and showing no stability in the air.

Peterboro, N. J. August 8, 1932

Edison, AVIATION

The article on morning flight by J. W. Miller in the issue of July 31 AVIATION is very interesting and particularly the analysis of light impact to a moving machine by making contact to a ball on a series of curved plates.

But the commentary on the theory of straight upward motion gives the doubt to the application. There is no possible doubt that nature of this sort is observable in bird flight. At least it has been the writer's privilege to see a small species of birds being pointed over a rapid on a perfectly level stretch of field, headed directly into a more or less steady wind and not less height for two or three minutes, yet with no visible effort except that of balancing.

To question Mr. Miller's explanation of a possible hovering or straight vertical motion, I beg to put it as follows.

If the position of the "CG" is far enough on advance of the center of pressure so that the forward component of the lift force along the path of flight just equals the drag, then obviously the path of flight will be a straight line. The horizontal flow of wind is downward, in, must be a "glide" is indicated in the first part of the hypothesis, (all our forces are hereby disposed of). And being a glide downward, on what basis can we possibly hope that the machine will "hover" (drift upward) by according a "wind moving hard enough" (WALTER L. WHEELER)

Eliminating the Airline Hangar

Reducing Investment in Buildings and Overhead by Mooring Weather-Proof Airplanes in the Open

By Archibald Black

Consulting Engineer, Garden City, N. Y.

In any place for operation of freight or passenger service as a comprehensive scale, the hangar has a very doubtful place, if indeed, it has any larger place there. If the airplane and airship are regarded as means of transport rather than as the type of working equipment, it becomes inconceivable to continue housing them when not in actual use, or undergoing inspection or overhaul. Steamships, railroad cars and power boats are probably the best comparisons which can be made with commercial aircraft. None of these are housed while out of service. Each is sufficiently waterproofed to withstand exposure to the weather and is left out in the open at all times.

Automobile a Poor Comparison

The automobile, to which the airplane is so frequently and so illogically compared, forms a very poor comparison. The automobile is essentially an individually owned machine. The airplane is essentially a vehicle which requires a complete

to be regularly housed while the field of which must be maintained and to leave them in the open if it is more convenient.

The airplane or airship, as compared with the automobile, is observed only while the passengers are getting in or out. If a freighter, its fleet will never be noticed and in neither case can its appearance be observed directly while it is in flight. Then comes, it is compared, which, however, a very considerable amount of space even if its wings are folded in every way it presents a conflict comparison with railroad equipment.

Hangars Few Formerly Shipyards

In the early days of aeronautics, as much repair work was constantly being done that the hangar was an essential. Every hangar was a shipyard and, indeed, most airplanes and airships were actually built in these hangars. In the case of large commercial airlines, such as are being developed today,



Here the flying boats of Aeromarine Airways are kept on shore for purposes of overhauling—an instructive demonstration of the sturdy, weather-proof qualities of these craft

operation in back of it, as do the locomotives. It does not adapt itself well to individual operation.

All passenger cars, and even many light trucks, are very highly finished on account of the close inspection to which they are subjected by those passing or passing them on the public streets. This close contact puts the passenger car, more or less, "on exhibition" at all times. It is something which is other than of vehicle today has to contend with. As a consequence, passenger cars are treated out with a finish and polish beyond comparison with any other type of vehicle. Exposure to the weather is very injurious to this finish, as the practice of housing all passenger cars and most trucks, when not in use, has become the custom.

In addition to this consideration, the automobile is used largely in office work. It is not necessary to leave it in the open at all times and the space which it occupies being comparatively small, the cost of housing it is a figure in no sense excessive. It is quite customary for trucks to be left out in the open in cases where their appearance is not of importance. In construction work these are often parked by the roadside, while at other times they are parked in construction yards without cover. In fact, the general rule applies

this condition is passing. So far as airplanes are concerned, it has already passed. The airplanes are built in regular factories and shipped, by air, rail or road, to the field. Shipping by air has become quite customary and it is probable the factory of the future will be located adjoining a field so that a large part of an airplane may be delivered in the shop. The field, in turn, equipped with several hangars and the major part of the maintenance work is usually centered in one of these buildings which becomes dependent on the weather. The evident tendency toward more and more mooring of the aircraft on buildings, in this way, is a most logical development.

It seems certain that, in time, all of the work will be thus centered at transportation fields, if only in the interest of efficiency. When this stage is reached, the older hangars will be used only for two purposes: first, storage of machines and, second, for the housing of machine undergoing inspection.

Estimation Possible by Elimination of Storage Hangars

If a comprehensive service is to be operated, storing of airplanes not in use will call for considerable investment in buildings. This becomes a serious matter with any transportation firm, both because of the additional capital economy

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The Aviation Country Club of Detroit
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TEN REASONS WHY

YOU SHOULD ATTEND
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 at **DETROIT, MICHIGAN**
DURING THE NATIONAL AIRPLANE RACES
OCTOBER 12th, 13th, and 14th, 1922

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- Two—The science of flight is your new medium of travel and transport. You must follow its progress if you are awake to the new era in communication.
- Three—The greatest number of new and modern planes ever assembled at a flying event will give you first hand information about the remarkable development of aircraft.
- Four—Representatives of the National Air Association, the Aero Club of America, and hundreds of prominent men are co-operating in the preliminary work of the Second National Aero Congress.
- Five—The Second National Aero Congress is called to create an American aeronautic association which shall represent your thought and echo your voice here and abroad, in all fields of business and in the halls of your Municipal, State and National legislatures.
- Six—You will have an active part in drafting the Constitution and By-Laws of this great representative organization. Your hand shall help write its policies, your efforts control its activities.
- Seven—An Advance Committee on Organization is preparing an elaborate program for your consideration and approval.
- Eight—Local committees are at work on details of the congress, to the end that your visit to Detroit may be replete with activities combining entertainment with convention business.
- Nine—Reunions of flying squadrons are being held. Meet your buddies at the Second National Aero Congress.
- Ten—Reduced railroad rates have been obtained. Your ticket agent can explain these details. Reservations can be arranged for you at Detroit's leading hotels where you will meet your co-workers and friends in aviation, and make acquaintances among the new arrivals in this constantly growing force, which believes, with Orville Wright, that "America, the birthplace of aviation, should lead the world in flight".

For further details write

The Advance Committee on Organization SECOND NATIONAL AERO CONGRESS

4612 Woodward Avenue,
 DETROIT, MICH.

307 Mills Building,
 WASHINGTON, D. C.

Kansas City Airport

An advertisement of months of efforts a report was received from Maj. Gen. Mason M. Fordrich, Chief of Air Service, stating it possible for Kansas City to be the site of a new aviation base in which the Air Service will place an airfield for the use of the Army, Post Office Department, Bureau of Aeronautics and National Guard and civilian enterprise. A site embracing 150 acres is being purchased and will be converted into a first class airport. The tract is situated a few hundred feet south of Blue Ridge Blvd. on the Eastport Rock Road (Old Santa Fe Trail), 5½ miles from the center of the city by airline.

The government proposes to use this field as an Air Service station for the testing and refreshing courses of the Reserve and National Guard as well as the Air Mail. The Quartermaster is now engaged in the erection of the Government buildings on the field. The site of the proposed field is being inspected by Maj. Gen. A. Badger of Ft. Crook, Neb., the Corps Area Air Officer, and Capt. C. H. Clark, Major, who is on the National Airway Board in Washington. Both these officers were well pleased and gave most favorable recommendations which have been since acted upon.

When this plan is realized Kansas City is the second city to benefit by the government's recognition, Boston being the first. The new airfield will be an airport of all flying at the field, and the flying route laid down by law will prevail. This will eliminate the possibility of the use of unsafe equipment by inexperienced pilots and the public will be assured of the safety ability and the surroundings of his visit.

The proposed airport is sponsored by the following group of men whose past experience and knowledge fit them particularly for this work: Eugene C. Crutcher, with the Flier Insurance Co., Frederick H. Harver, with Ford Harvey, Inc., Robert R. Loebe, with Geo. B. Loebe, and also Messrs. F. W. Wolfe, with The Kansas Manufacturing Co., Hanson Tomlinson, with the J. C. Nichols Van Co. These men have committed themselves to undertake to meet the government's requirements by providing the necessary field, arranging for its financing, superintending its construction and commercial activities.

The efforts of these men to establish a Kansas City air terminal and the financing plan to make it possible have been indicated by the Chamber of Commerce and the Flying Club of Kansas City.

Two companies are provided for a land company to purchase and hold title to a 150-acre tract, and an operating company to lease the ground from the land company and operate the flying field in conjunction with the United States government.

During the last few weeks fifty-eight Kansas City business firms and individuals have subscribed \$100,000 worth of stock to both the land company and the operating company of the Kansas City air terminal association, and it is confidently expected that the total capitalization of \$150,000 required for the project will be subscribed before long.

Maj. Wolfe is particularly equipped successfully to direct a project of this kind. Having served in five of the Cavalry Regiments in April, 1916, he was called into the Service of the United States Army in July of that year. In the course of his various duties he assisted in laying out and superintended the construction of Bluebird Field at Monticello, N. Y., the first government school in the East and the second in the United States. Many of his ideas, as worked out at Bluebird, were embodied in the construction of the twenty-four aviation schools built throughout the country during the war.

Plans have been made for various commercial activities which naturally grew up around a large Government Air Base, namely: Provisional Service, personnel facilities, storage, service and repair for shops, the lease of a portion of the property to a school of aviation where the license may be taught by it. All the activities will enjoy the advantages of central control and efficient supervision.

It is furthermore expected that the Flying Club of Kansas City will erect a Club House on the field.

Recent Developments and Outstanding Problems

N.A.C.A. Report No. 132

The report, by Franklin L. Root, which is the concluding number of a series of reports on aerometric instruments, discusses briefly some of the more recent developments in aerodynamic instrument design and mentions some of the outstanding problems.

The different types of instruments are discussed as far as possible in the order in which they are discussed in the preceding reports. Among the instruments considered are pressure altimeters and barographs, a thermocouple altimeter which automatically corrects for the effect of air temperature on altitude readings, rate of climb recorder, a combined altimeter and rate of climb recorder, low speed air-speed indicator for light-aircraft craft, ground speed indicator, anemometer, wind indicator, aerodynamic instruments, also special instruments relating to the altitude effect of air-speed indicators, the properties of static discharges, flow-line tubes and barometric laws and tabulations for recording instruments.

Among the outstanding problems are mentioned the need of recording instruments for long distance flights and for flying at high altitudes and landing in line, the improvement of oxygen instruments for high altitude flying, and the attainment of greater reliability and precision in all aerometric instruments.

The Bristol "Lucifer" Engine

The Bristol "Lucifer" engine, which is illustrated herewith in line view, represents the 350-hp. model, and also bears the British air engine. In this connection, and also because the



Front view of the 350 hp Bristol "Lucifer" 2 cyl. radial aero-engine

"Lucifer" is the only British air-cooled radial engine of medium horsepower which has passed the experimental stage. The following particulars will be found of interest.

SPECIFICATIONS OF THE BRISTOL "LUCIFER" ENGINE	
No. and type of cylinders	2, radial
Bore	3 1/2 in.
Stroke	4 1/2 in.
Displacement	400 c.i. in.
Compression	16 to 1
Rated hp. (1000 r.p.m.)	350
Maximum hp. (1400 r.p.m.)	400
Full consumption	0.610 lb. per hp. hr.
Specific consumption	0.55 lb. per hp. hr.
Weight and oil and fuel	175 lb. including tank
Weight for 4-1/2 in.	100 lb. 10 in. tank

The Improved Liberty Engine

The Liberty 300-hp. aircraft engine which was brought out in this country during the war was designed in England as an engine for a standardized aircraft engine that would be placed in quantity production at a time when the air forces of the country were expanded from a negligible quantity to a vast host of planes. At the time the Liberty engine had the unqualified approval of designers and manufacturers in this country, who maintained that it was the best product available in aircraft engines. During the war it was used in an extensive manner, and in numerous and varied events.

In the light of later experience, it has developed that the Liberty was not, as it was at first claimed for it, the perfect aircraft engine. It had many inherent faults which resulted in unreliability and caused many failures and forced landings. On the other hand the many good qualities of the engine were not so much as to stimulate every effort to correct existing deficiencies rather than to turn to new designs. To this the Bureau of Aeronautics, Navy Department looked the problem, and by studying and the greatest difficulties and working down through to minor faults was able by progressive steps to place at the service of the Naval Air Branches engines capable of air duty that might be expected of a work of magnitude of dependability and endurance.

At the time of the American Navy had some 4000 Liberty engines representing an outlay of approximately \$10,000,000 of government funds. The engines were subject to many unexpected trouble which amounted forced landings of frequent occurrence and caused accidents.

Among the difficulties experienced were trouble with breaking of timing pins, excessive oil consumption and feathering of propellers, generally at low speeds. Cylinder pistons also caused considerable trouble from cracking, allowing loss of water, which produced failures were another cause of worry. For a time it appeared as if the very design of the Liberty engine was at fault.

The most serious fault encountered was with gear failures and it was in this that the naval authorities gave first consideration. A redesign of the timing gear was effected and the trouble was completely eliminated. Next the pistons were modified by cutting a groove across the edge of the lower piston ring groove and dividing oil return holes in the side of the piston. With this change the piston ring trouble was eliminated as it prevented oil from getting into the combustion chamber. It also reduced the oil consumption and as a result of this the oil pressure was raised. With increased oil pressure

the bearing life was lengthened and this lengthened the time between overhauls.

In order to eliminate propeller failures, a flexible drive was substituted and up to the present time has proved satisfactory. To prevent the cracking of water jackets, a water-jacket strip has been added across the top of the cylinder on the inside of the water jacket; this has given excellent results.

The Liberty engine ignition system had never been entirely satisfactory. Recent changes carried out by the Bureau of Aeronautics include the substitution of a 12-volt ignition system in place of the 6-volt system, whereby the battery cannot discharge so rapidly. It also allows much easier starting.

With the above changes in effect, the Liberty engine has been in use in planes in the operating squadrons throughout the past winter and spring with excellent results. During the recent flying of twelve F5A airplanes from Ellington Field to Pensacola the performance of the engine indicated the changes made due with the exception of one forced landing due to a leaky water jacket, the engine came through twenty-five hours of flying without the slightest trouble.

In a report to the Bureau of Aeronautics the Commander of the Air Squadron, Ellington Field, page high tribute to the work that has been done in this connection.

In general all the modifications to the standard Liberty have proved of great value," says the report. "The increase in oil pressure has probably doubled the life of the motor." The use of heavy duty has eliminated trouble from the motor provided the commanding officer will give their power plants the necessary wear and tear. The Liberty engine has been in use in planes in the operating squadrons throughout the past winter and spring with excellent results. During the recent flying of twelve F5A airplanes from Ellington Field to Pensacola the performance of the engine indicated the changes made due with the exception of one forced landing due to a leaky water jacket, the engine came through twenty-five hours of flying without the slightest trouble.

As a general result of the winter work, the Commander of the Air Squadron is convinced that the improvement in material is such that the time is fast approaching when forced landings and delays starting can be largely eliminated on material but the responsibility therefore can be placed squarely up to the operating personnel.

The alterations and changes in the Liberty engine which have been effected, were under the direction of Lieut. Col. George H. M. Krenn, and Lieut. Col. Landon of the Bureau of Aeronautics and others commanding their work in this connection have been addressed to them by the Chief of the Bureau.



The Kistner K-11 commercial airplane (250 hp. Le Rhône engine)—a quarter side view showing details of engine (main top)

444

Foreign News

Amelia—Information as to the present condition of mail service in Australia, sent by the Commonwealth Director of Aviation, states that there are now about only landing grounds—emergency and otherwise—prepared in different parts of the country. The aim is to have an emergency landing ground every fifteen miles of surveyed routes. These grounds are about 1500 yd. in extent, and are marked in the center by a large outline such as white. When more funds are available it is intended to make supplies of gasoline and to employ aircrews at each of the principal landing places. The government has last year for civil aviation was \$250,000 and it has been indicated that next financial year a larger sum will be placed on the situation.

At present there is only one aerial mail service in operation, between Queensland and Derby on the continent coast of Western Australia, the distance between the two places being about 1500 miles with five intermediate stops. This service is claimed to be 100 per cent efficient, every trip during the past five months having been carried out according to schedule. Contracts have been let for services between Queensland and Glenora in Queensland, and between Sydney and Adelaide, and Sydney and Brisbane. It is expected that the Queensland service will be commenced next August, and the other services when additional machines arrive from England.

When all the services already indicated for are in operation, the extent of country to be traversed will be 3000 miles.

Denmark—The Danske Luftfartsselskab (Danish Air Transport Co.) having obtained official approval for one-half of a possible deficit, is arranging extensive traffic connections with Central Europe. The company, in its late contract and in exchange for the government's promise of assistance.

The daily service to Hamburg can take 100 kg. of freight or three passengers, but the route is established only with the idea of taking freight and mail. Germany will furnish three and Denmark three of the machines used on the Copenhagen-Hamburg line. The Danish planes are reliable and improved DeHavillands.

Italy—According to London newspapers, Signor M. Gi. Pavesi, the Italian aviator, has designed a 5 hp. single-seater airplane, which it is claimed will be the smallest flying machine in the world. The machine has a single lifting wing, designed from ideas received on machine building, and also is said to be foldable, so that the machine can be packed in space a small shed. The airplane is now being built in Rome, and is to be fitted with an engine designed to facilitate mass production.

Bombing Machine in California—A crowd, variously estimated between 25,000 and 50,000 people, witnessed the bombing of the schooner Nevada off Fort Crockett, California, Tex., recently by the First Pursuit Group. The bombing was done with demolition of fire, since the first bombs obtained results. The machine used in the type with which the ships are equipped. Even so, the last attack left the Nevada on fire in the hold, in front into flames upon the withdrawal of the planes. The opening attack was made with two 5-lb. fragmentation of flying bombs dropping four 25-pounders and getting a very creditable percentage of hits on deck, in spite of the fact that the machine stands for five bombs were released. The bombs were followed by a formation of five COBAs from Kelly Field dropping fuel loads of 25-pounds.

It was obvious to any observer that, if bombs of the type actually used against craft of this type had been employed here, every man in the fleet would have been with open to the sea after the COBAs were half through. This attack was followed by two formations of REVs dropping 4 bombs each, 3 of the planes remaining to attack with machine guns.

The First Pursuit Group expects to have more of this sort of thing at Redondo Beach, the second objective period of Lake St. Clair offering a convenient location for either practice or demonstration.

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